

DESIGN OPTIMIZATION AND CFD ANALYSIS OF CAR USING ACTIVE MOUNTING TO REDUCE DREG COEFFICIENT AND REDUCE LIFT TO **SLEEP OF CAR AT HIGHER SPEED**

RAJYAGURU PRANAV ¹, TANDEL HITESH ², KISHAN VAGHELA ³,

123 student-Vadodara Institute Of Engineering -Vadodara-India

Abstract: An analytical study on air flow which effects and resulting dynamics on the car is presented. The research deal with Computational Fluid Dynamic analysis and simulation to maximize down force and minimize drag during high speed of the car. Using ANSYS FLUNT software and mentoring provided by ANSYS, the results employs efficient discretization techniques and real loading conditions to study down force on rear wing of the car with drag generated by all active mounted surfaces. Wing and external surface under high velocity runs of the car are presented. Optimization of wing direct angle of attack and geometry modifications on surfaces of the car are performed to enhance down force and reduce drag for higher degree of stability and to control during operation. Moreover to ensure more stability active aerodynamic spoiler which can adjust its height and its angle of attack according to speed. So the aim of research to find best height at which maximum down force and air flow separate ,so that car body can get high level of stability . And another factor is angle of attack at which highest value of down force we get .

Keywords: ACTIVE AERODYNAMICS, SPOILER, DOWN FORCE, STABILITY, ANSYS (FLUENT), CATIA V6

1. INTRODUCTION

1.1 Introduction to active aerodynamics

Active aerodynamics is a combination of a various -stage adjustable rear spoiler. The spoiler lip is protected. In stage two (Speed), after 120 km/h, rear spoiler are partially extended. This helps to ensure a considerable level of stability, a low drag coefficient and enables a high top speed rear spoilers are now fully extended. Also, in this position, the rear spoiler is tilted by up to 15 degrees.

1.2 COMPUTIONAL FLUID DYNAMICS

Computational fluid dynamics is the transformation of a situation to the mathematical model with governing equation. This whole equation are looped with other equations which are depend on another equations and solved together however, this calculation is very tough by manual source so it is calculated by metleb, ansys, simscale and various cfd soft wear. In background in cfd the equations of mass, momentum and energy also for drag and lift called navier stock equation . so that cfd is called combination of computer science, mathematics, physics in our research we don't require any heat transfer or any thermal turm we only interested in air streamlines .

1.2 MODELLING

The 3-d model is done using blue print of Porsche 911 turbo .Automobile body have lots of sharp curves and edges. so that it is very difficult to make solid model. By using of surface modeling automotive body can easily done and results will be perfect and can be implemented in real word and also used by automobile company.

SOFTWERE USD IN MODELING

- 1. CREO 2.0
- 2. CATIA V5
- 3. NX NASTERAN



FIG-1 SURFACE MODEL OF CAR

2. MESHING

. For this the 3D model with enclosure was imported, then the enclosure was given named selections into various parts like body, wall, inlet and outlet and the required meshing conditions were applied and meshing was done on the imported geometry. Meshing is an important step of whole analysis because results are mostly depends on number of elements and node as the number of elements are high than order of accuracy is high.

2.1 Meshing Size

ELEMENTS	3667923				
Use NODES	659019				
Adv. Size Fun .	On: Proximity and Curvature				
Minimum Size	17.23mm				
Maximum Size	500mm				
TABLE-1 MESH SIZING					



FIG-2 MESH GENERATION



3. VIRTUAL TUNNEL GENERATION Virtual

Tunnel is one enclose in which car model is put in mid position . In real condition car is moving however, in ANSYS car is steady and air is pass through tunnel



FIG -3 WIND TUNNEL

4. ANALYSIS

The analysis of the model with enclosure was done in ANSYS FLUENT 16.2. For this at first the mesh was checked and after the approval of mesh various analysis parameters like models, materials and boundary conditions were set

4.1. Models

The model used for analysis was K-epsilon.

4.2 BOUNDRY CONDITION

Inlet type	Velocity inlet		
Outlet type	Flow-split outlet		
Top and side faces	Slip shear stress specification		
Symmetry face	Symmetry		
Ground face	Wall (with the same velocity as the		
	flow)		
Car faces	Wall		
Pressure	Atmospheric		
Turbulence Intensity	3 %		
Turbulence Specification	Intensity + length scale		
Turbulent Length Scale	10 mm		
Turbulent Velocity Scale	1 m/s		
Velocity	40 m/s		

TABLE -3 BOUNDARY CONDITION

5. RESULTS AND DISCUSSIONS

Results are last stage but an important one .geometry of our research is too big in volume so the comperation from all numerical value is become too complex and time consuming . so we try to do this based on color contour however it is the second form of numerical values

In our research we had target to complete analysis at five angle of attack and at three speed so whole procedure we can not present here and we presented limited and important results.

Using ANSYS we complete analysis in two parts

(1) analysis of car body without spoiler at various speed

(2) analysis of car body using spoiler at various angle of attack and at various height

5.1 PRESSURE COUNTER AT 50M/S

In automotive analysis due to large number of elements and volume it is very complex to take every mathematical data . To save time and reduce complexibility comparison of color counter is a way to compare down force



FIG-4 EFFECT OF BACK PRESSURE AND IRREGULAR STREAM LINES



FIG-5 EFFECT OF DIFFUSER



FIG-6 STREAM LINES OVER CAR BODY





FIG-7 PRESSURE CONTOUR AT 50M/S WITHOUT SPOILER



FIG-8 PRESSURE CONTOUR AT 70M/S



FIG-9 GRAPH OF RESULT OF DREG AND LIFT AT 35M/S,50M/S,70M//S









FIG-11 GRAPH OF COMPARISION OF GRAG AND DOWN FORCE

6. CONCLUSION

In our research we had lots of analysis about 20-30 so it become very complex to conclude bast angle attack for any speed moreover speed such as 70-80m/s is very higher speed so that analysis should be very exact. At some stage sych as 30- 40 m/s we required only slip don't require too much down force . in addition to that we cant focuse only on down force because at some value down force is high but drag is also high that is why we cannot use that angal at perticular angal.

At the very high speed down force is very crusial turm also it required to check the streamlines , that should not creat the back pressure and negative pressure . cfd is dafinately a best way to see aerodynamics however it is too complex and too much calculation is required with higher order of accuracy so that it can be applied

Although the main goal was to pick the highest value of Lift forces to Drag forces ratio, the absolute values of the lift forces and drag forces also had to be considered For example, the actual lift was the least of all angle-of- attack orientations. Hence the end goal was to identify the highest lift possible and then utilize the ratios as a method of optimization (see Table 3)

Here in table some of the best results are shown here we can not show all the combination here some of best result as per our goal is shown here . on this results active spoiler is designed by use of micro controller .



www.irjet.net

Lift:DragRatio	5(degree)	6(degree)	8(degree)	13
				degree
150km/hr	8.92	8.87	9.02	9.12
200km/hr	9.06	8.31	7.76	8.91
250km/hr	7.8	7.2	8.32	8.94

TABLE -4 DATA FOR ACTICE SPOILER

7. REFERENCE

1. John J. Bertin, "Aerodynamics for Engineers", Prentice Hall; 5th edition, New Jersey, June 2008

2. Jiyuan Tu, Guan Heng Yeoh and Chaoqun Liu, "Computational Fluid Dynamics: A Practical Approach", Butterworth-Heinemann; 1st edition, Burlington, MA, November 2007

3. Oleg Zikanov, "Essential Computational Fluid Dynamics", John Wiley & Sons, Inc. Hoboken, New Jersey, March 2010

4. C. H. K. Williamson, "Three Dimensional Vortex Dynamics in Bluff Body Wakes", Experimental Thermal and Fluid Science, Volume 12, February 1996, p. 150-168

5. 1. Wolf-Heinrich Hucho, "Aerodynamics of Road Vehicles: From Fluid Mechanics to Vehicle Engineering", Society of Automotive Engineers Inc; 4th edition, Warrendale, Pa, February 1998

6. Website: <u>http://autospeed.com/cms/title_Aero-Testing-Part4/A_108676/article.html</u>

7. Marco Lanfrit, "Best practice guidelines for handlingAutomotive External Aerodynamics with FLUENT", Fluent Deutschland GmbH, 64295 Darmstadt/Germany, February 2005

8. W. Seibert "CFD in Aerodynamic Design Process of Road and Race Cars", Fluent Deutschland GmbH, FLUENT Technical Notes TN155, Presented at European Automotive Congress, Bratislava, Slovakia, June 18-20 2001

9. Klaus Gersten, E. Krause, H. Jr. Oertel, C. Mayes "Boundary-Layer Theory", Herrmann Schlichting, 8th Edition, Springer 2004 10. W. Seibert, M. Lanfrit, B. Hupertz and L. Krüger "A Best- Practice for High Resolution Aerodynamic Simulation around a Production Car Shape" 4th MIRA International Vehicle Aerodynamics Conference, Warwick, UK, October 16-17, 2002

11. "FLUENT 6.0 User's Guide - Volume 1", ANSYS Inc., New York, December 2001.

12. "ANSYS FLUENT 12.0/12.1 in Workbench User's Guide", ANSYS Inc., New York, October 2011

13. "ANSYS FLUENT 12.0 Theory Guide", ANSYS Inc., New York, April 2009

14. Masaru KOIKE, Tsunehisa NAGAYOSHI and Naoki HAMAMOTO, "Research on Aerodynamic Drag Reduction by Vortex Generator", Mitsubishi Motors Technical Reviews, Tokyo, Japan, 2004

15. M. Rouméas, P. Gilliéron & A. Kourta "Drag Reduction by Flow Separation Control on a Car After Body", Int. J. Numer. Methods Fluids 60, 2008. ISSN 0271-2091

16. A. Kourta, P. Gilliéron "Impact of the Automotive Aerodynamic Control on the Economic Issues", Journal of Applied Fluid Mechanics , Vol. 2, No. 2, pp. 69-75, 2009, ISSN 1735-3645

17. Byoung-Kwon Lee, MD "Computational Fluid Dynamics in Cardiovascular Disease" Korean Circulation Journal 41(8): 423–430, August, 2011

18. Website: <u>http://bugattipage.com/ride.htm</u>

19. Website: http://www.nasa.gov/audience/forstudents/5-8/features/what-isaerodynamics-58.html

20. Website: http://www.grc.nasa.gov/WWW/K-12/airplane/boundlay.html